

Outline



- > Air cycles
 - What are they?
 - ➤ Why use P-V and T-s diagrams?
- ➤ Using P-V and T-s diagrams for air cycles
 - > Seeing heat, work and KE
 - > Constant P and V processes
 - > Inferring efficiencies
 - > Compression & expansion component efficiencies
 - > Correspondence between processes on P-V and T-s
 - > Hints & tricks

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Air-cycles - what are they?



- Mostly we'll analyze "air cycles" working fluid is air (or other ideal gas) with constant properties
- \triangleright Changes in C_P, M_i, γ , etc. due to changes in composition, temperature, etc. are neglected; simplifies analysis, leads to simple analytic expressions for efficiency, power, etc.
- ➤ Later we'll analyze "fuel-air cycles" (using GASEQ) where the changes in C_P, M_i, γ, etc. due to changes in composition, temperature, etc. are considered
- ➤ "Fuel-air cycles" still can't account for slow burn, heat loss, etc. since it's still a thermodynamic analysis that tells us nothing about reaction rates, burning velocities, etc.
- P-V and T-s diagrams provide a clear visual representation of cycles

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Why use P-V diagrams?



- Pressure (P) vs. time and cylinder volume (V) vs. time are easily measured in reciprocating-piston engine experiments
- ▶ ∫ PdV = work
- ▶ ∫ PdV over whole cycle
 = net work transfer + net change in KE + net change in PE
 = net heat transfer
- Heat addition is usually modeled as constant P or constant V, so show as straight lines on P-V diagram
- ➤ Note that V on P-V diagram is cylinder volume (units m³)
 - ➤ NOT a property of the gas (it's a property of the cylinder!)
 - > Specific volume (v = V/m) (units m³/kg) IS a property of the gas
 - ➤ We need to use V not v since cycle work = ∫ PdV
 - Can't use v since mass in the cylinder (m) changes during intake / blowdown / exhaust

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Why use T-s diagrams?



- Idealized compression & expansion modeled as constant S
 - \triangleright dS ≥ δQ/T (an important consequence of the 2nd Law)
 - For adiabatic process $\delta Q = 0$, for reversible = (not >) sign applies, thus dS = 0
 - ➤ Note that dS = 0 still allows for any amount of work transfer (W) to or from the gas
 - ➤ Note 2^{nd} law does NOT lead to $dS \ge \delta Q/T \delta W/T$ because there is no entropy change due to work transfer
- ➤ For reversible process, ∫TdS = Q, thus area under T-S curves show amount of heat transferred
- > T-S diagrams show the consequences of non-ideal compression or expansion (dS > 0)
- > For ideal gases, ΔT ~ heat transfer or work transfer or ΔKE
- Efficiency can be determined by breaking any cycle into Carnot-cycle "strips," each strip (i) having η_{th.i} = 1 - T_{L.i}/T_{H.i}

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T-S & P-V for control mass: work, heat & KE^{USC}Viterbi

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➤ For an ideal gas with constant C<sub>P</sub> & C<sub>V</sub>:
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$$\Delta h = C_p \Delta T$$
, $\Delta u = C_v \Delta T$

1st Law for a control mass with $\Delta PE = 0$ (OK assumption for IC engines) $dE = d(U + KE) = d[m(u + KE)] = \delta q - \delta w$ (using "KE" as in "Kinetic Energy" rather than "u²/2" avoid confusion with u

(using "KE" as in "Kinetic Energy" rather than "u²/2" avoid confusion with ι (internal energy))

q = Q/m; w = W/m (heat or work transfer per unit mass)

 \triangleright If no work transfer ($\delta w = 0$) or KE change ($\Delta KE = 0$),

$$du = C_v dT = \delta q \Rightarrow q_{1\rightarrow 2} = C_v (T_2 - T_1)$$

ightharpoonup If no heat transfer ($\delta q = 0$) or ΔKE

$$du = C_v dT = \delta w \Rightarrow w_{1\rightarrow 2} = -C_v (T_2 - T_1)$$

 \rightarrow If no work or heat transfer ($\delta q = \delta w = 0$)

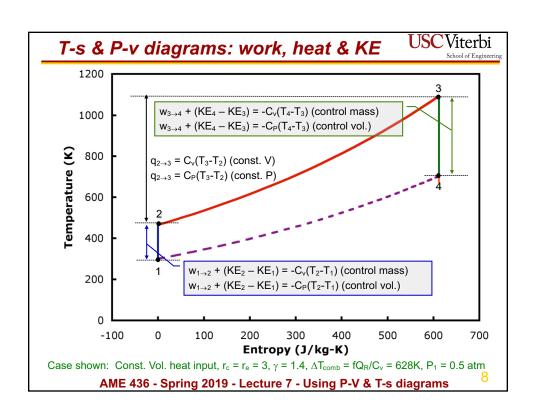
$$\Delta KE = KE_2 - KE_1 = -C_v(T_2 - T_1)$$

- \Rightarrow For a control mass containing an ideal gas with constant C_v , $\Delta T \sim$ heat transfer work transfer ΔKE
- ➤ Note that the 2nd law was not invoked, thus the above statements are true for any process, reversible or irreversible

T-s & P-v for control volume: work, heat & KESCViterbi

- > 1st Law for a control volume, steady flow, with ΔPE = 0 $0 = \dot{Q} - \dot{W} + \dot{m} \left[(h_{in} - h_{out}) + \left(KE_{in} - KE_{out} \right) \right]$
- For ideal gas with constant C_P , $dh = C_P dT \Rightarrow h_2 h_1 = C_P (T_2 T_1)$
- ► If 2 = outlet, 1 = inlet, and noting $\dot{Q}/\dot{m} = q_{1\to 2}; \dot{W}/\dot{m} = w_{1\to 2}$ $h_2 - h_1 = q_{1\to 2} - w_{1\to 2} - (KE_2 - KE_1)$
- ► If no work transfer KE change $(w_{1\rightarrow 2} = \Delta KE = 0)$ $dh = C_P dT = dq \Rightarrow q_{1\rightarrow 2} = C_P (T_2 - T_1)$
- ► If no heat transfer (dq = 0) or Δ KE dh = C_PdT = δ w \Rightarrow w_{1→2} = -C_P(T₂ - T₁)
- ► If no work or heat transfer $\Delta KE = KE_2 - KE_1 = -C_P(T_2 - T_1)$
- \Rightarrow For a control volume containing an ideal gas with constant $C_{\text{P}},$ ΔT ~ heat transfer work transfer ΔKE
 - (Same statement as control mass with C_P replacing C_v)
- Again true for any process, reversible or irreversible

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T-s & P-v diagrams: work, heat, KE & PESC Viterbi

 \succ How do I know that work shown on the P-V (via $\int PdV$) diagram is the same as that shown (via $C_v\Delta T$) on the T-s diagram? As an example, for isentropic compression (PV^{γ} = constant)

$$\begin{split} W_{2\to3} &= \int_{2}^{3} P \, dV = \int_{2}^{3} \frac{P_{2} V_{2}^{\gamma}}{V^{\gamma}} \, dV = -\frac{P_{2} V_{2}^{\gamma}}{\gamma - 1} \left(\frac{1}{V_{3}^{\gamma - 1}} - \frac{1}{V_{2}^{\gamma - 1}} \right) \\ &= -\frac{P_{2} V_{2}^{1}}{\gamma - 1} \left(\frac{V_{2}^{\gamma - 1}}{V_{3}^{\gamma - 1}} - \frac{V_{2}^{\gamma - 1}}{V_{2}^{\gamma - 1}} \right) = -\frac{mRT_{2}}{\gamma - 1} \left(\left(\frac{V_{2}}{V_{3}} \right)^{\gamma - 1} - 1 \right) \\ &= -mC_{v} T_{2} \left(\left(\left(\frac{T_{3}}{T_{2}} \right)^{\frac{1}{\gamma - 1}} \right)^{\gamma - 1} - 1 \right) = -mC_{v} T_{2} \left(\frac{T_{3}}{T_{2}} - 1 \right) = -mC_{v} \left(T_{3} - T_{2} \right) \end{split}$$

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T-s & P-v diagrams: work, heat, KE & PE



➤ Going back to the 1st Law again

$$dE = \delta Q - \delta W \Rightarrow \oint dE = \oint \delta Q - \oint \delta W$$

➤ Around a closed path, since E = U + KE + PE; since U is a property of the system, ∫dU = 0, thus around a closed path, i.e. a complete thermodynamic cycle (neglecting PE again)

$$\oint \delta Q = \oint \delta W + \Delta K E$$

But wait - does this mean that the thermal efficiency

$$\eta_{th} = \frac{\oint \delta W + \Delta KE}{\oint \delta Q} = 1?$$

No, the definition of thermal efficiency is

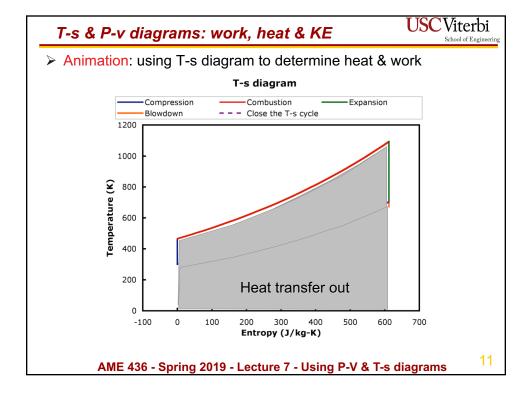
$$\eta_{th} = \frac{\text{What you get}}{\text{What you pay for}} = \frac{\oint \delta W + \Delta KE}{\text{Heat in}} = \frac{\text{Heat in - heat out}}{\text{Heat in}} < 1$$

For a reversible process, $\delta Q = TdS$, thus $\delta q = Tds$ and

$$\oint T ds = \oint \delta q = \oint \delta w + \Delta KE / m$$

> Thus, for a reversible process, the area inside a cycle on a T-s diagram is equal to both (net work transfer + net KE) and net heat transfer

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Constant P and V curves

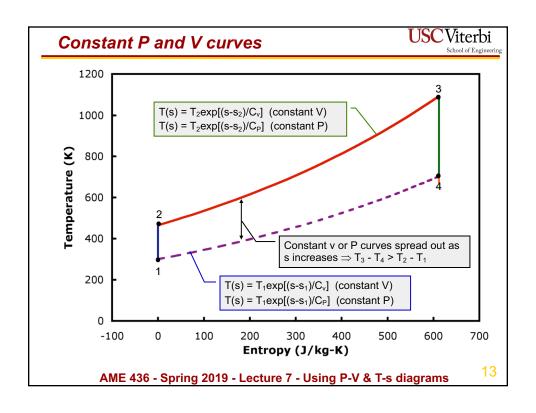


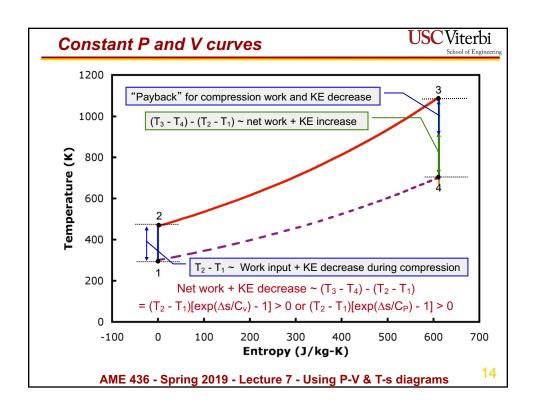
For ideal gas with constant specific heats, review of thermodynamics (1st lecture) showed that

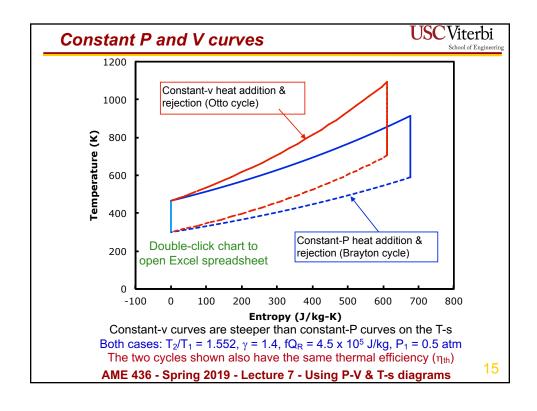
$$s_2 - s_1 = C_P \ln\left(\frac{T_2}{T_1}\right) - R \ln\left(\frac{P_2}{P_1}\right); \ s_2 - s_1 = C_V \ln\left(\frac{T_2}{T_1}\right) + R \ln\left(\frac{V_2}{V_1}\right)$$

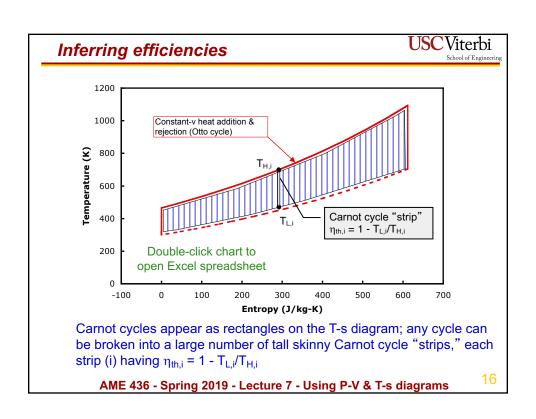
- ightarrow If P = constant, $ln(P_2/P_1) = 0 \Rightarrow T_2 = T_1 exp[(s_2-s_1)/C_P]$
- ➤ If V = constant, $In(V_2/V_1) = 0 \Rightarrow T_2 = T_1 exp[(s_2-s_1)/C_v]$ ⇒ constant P or V curves are exponentials on a T-S diagram
- Since constant P or V curves are exponentials, as S increases, the ΔT between two constant-P or constant-V curves increases; this ensures that compression work is less than expansion work for ideal Otto (const. V) or Brayton (const. P) cycles
- ightharpoonup Since $C_P = C_V + R$, $C_P > C_V$ or $1/C_P < 1/C_V$, constant V curves rise faster than constant P curves on a T-s diagram
- Constant P or constant V lines cannot cross (unless they correspond to cycles with different C_P or C_V)

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Compression & expansion efficiency



- > If irreversible compression or expansion, dS > $\delta Q/T$; if still adiabatic ($\delta Q = 0$) then dS > 0
- \succ Causes more work input (more ΔT) during compression, less work output (less ΔT) during expansion
- \triangleright Define compression efficiency η_{comp} & expansion efficiency η_{exp}

 $\eta_{comp} = \frac{\text{Reversible adiabatic work input for given V or P ratio}}{\text{Actual work input required for same V or P ratio}}$

$$=\frac{-C_{V}[T_{1}(V_{1}/V_{2})^{\gamma-1}-T_{1}]}{-C_{V}(T_{2}-T_{1})}=\frac{(V_{1}/V_{2})^{\gamma-1}-1}{T_{2}/T_{1}-1}$$

(control mass, specified volume ratio)

 $\eta_{\text{exp}} = \frac{\text{Actual work output for given V or P ratio}}{\text{Reversible adiabatic work output for same V or P ratio}}$ $= \frac{-C_V(T_2 - T_1)}{-C_V[T_1(V_1/V_2)^{\gamma - 1} - T_1]} = \frac{T_2/T_1 - 1}{(V_1/V_2)^{\gamma - 1} - 1}$

(control mass, specified volume ratio)

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Compression & expansion efficiency



$$\eta_{comp} = \frac{-C_{V}[T_{1}(P_{2}/P_{1})^{(\gamma-1)/\gamma} - T_{1}]}{-C_{V}(T_{2} - T_{1})} = \frac{(P_{2}/P_{1})^{(\gamma-1)/\gamma} - 1}{T_{2}/T_{1} - 1}$$

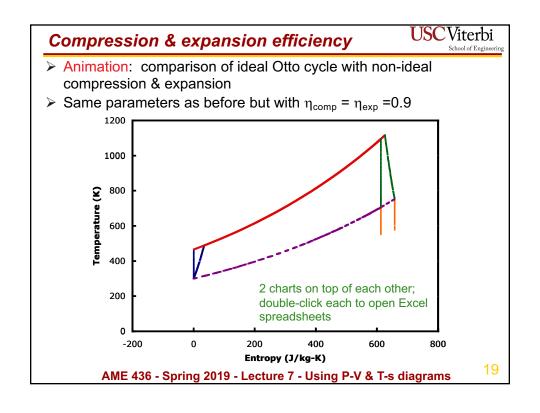
(control mass, specified pressure ratio)

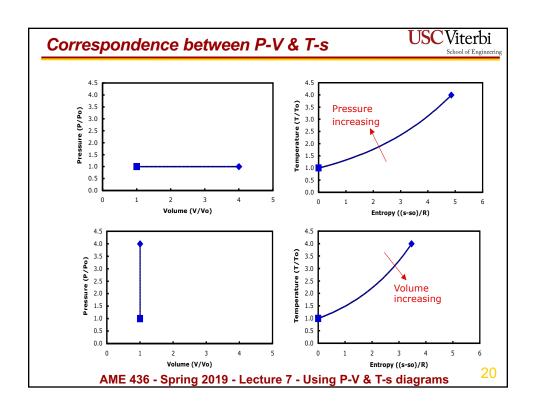
$$\eta_{\text{exp}} = \frac{-C_V (T_2 - T_1)}{-C_V [T_1 (P_2 / P_1)^{(\gamma - 1)/\gamma} - T_1]} = \frac{T_2 / T_1 - 1}{(P_2 / P_1)^{(\gamma - 1)/\gamma} - 1}$$

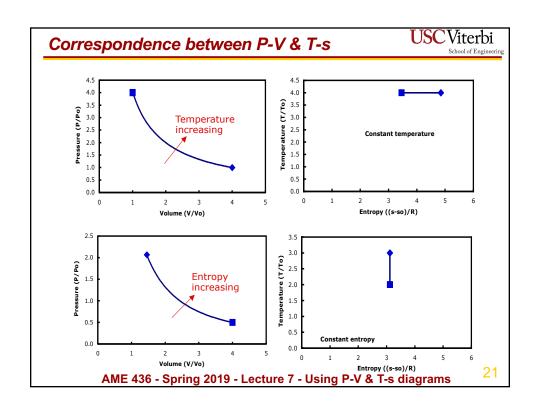
(control mass, specified pressure ratio)

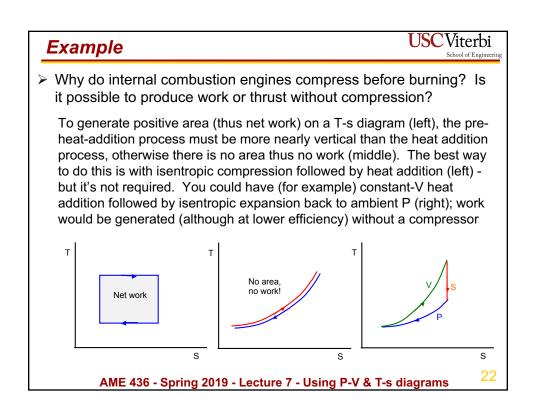
- ➤ Control volume: replace u (internal energy) with h (enthalpy) and thus replace C_v with C_P, but it cancels out so definitions are same
- ➤ These relations give us a means to quantify the efficiency of an engine component (e.g. compressor, turbine, ...) or process (compression, expansion) as opposed to the whole cycle

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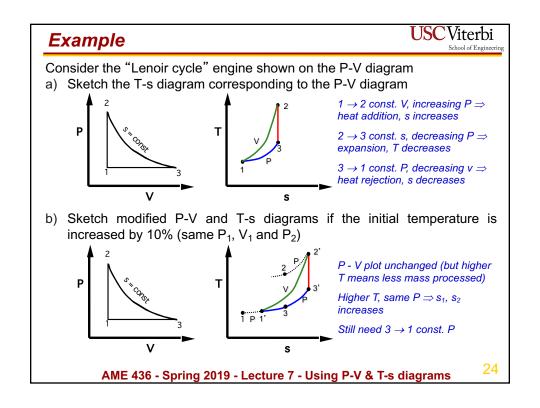


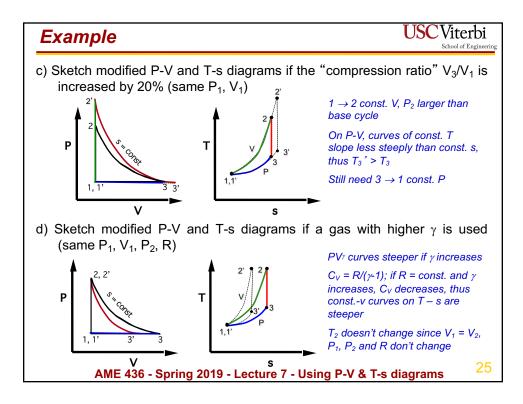






Example ■ USC Viterbing School of Engineering School of Enginee





Using T-s and P-v diagrams - summary USC Viterbi

- > Thermodynamic cycles as they occur in IC engines are often approximated as a series of processes occurring in an ideal gas
- > T-S and P-V diagrams are very useful for inferring how changes in a cycle affect efficiency, power, peak P & T, etc.
- ightharpoonup The ΔT (on T-S diagrams) and areas (both T-S & P-V) are very useful for inferring heat & work transfers

Using T-s and P-v diagrams - summary $\overline{\mathrm{USC}}$



- ➤ Each process (curve or line) on a T-s or P-v diagram has 3 parts
 ➤ An initial state
 - > A process (const. P, V, T, S, area, etc.)
 - ➤ A final state
 - » For compression and expansion processes in piston engines, a specified V (i.e., a particular compression or expansion ratio)
 - » For compressors in propulsion cycles, a specified pressure ratio
 - » For turbines in propulsion cycles, a specified temperature that makes the work output from the turbine equal the work required to drive compressor and/or fan
 - » For diffusers in propulsion cycles, a specified Mach number
 - » For nozzles in propulsion cycles, the pressure after expansion (usually ambient pressure)
 - » For heat addition processes, either a specified heat input = \int Tds (i.e. a mixture having a specified FAR and Q_R) thus a given area on the T-s diagram, or a specified temperature (i.e. for temperature limited turbines in propulsion cycles)
 - » The constant P and constant V exponential curves on the T-S diagram are very useful for determining end states

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Using T-s and P-V diagrams - summary



- > Three or more processes combine to make a complete cycle
- ➤ When drawing P-V or T-S diagrams, ask yourself
 - ➤ What is the P, V, T and S of the initial state? Is it different from the baseline case?
 - > For each subsequent process
 - » What is the process? Is it the same as the baseline cycle, or does it change from (for example) reversible to irreversible compression or expansion? Does it change from (for example) constant pressure heat addition to heat addition with pressure losses?
 - » When the process is over? Is the target a specified pressure, volume, temperature, heat input, work output, etc.?
 - » In gas turbine cycles, be sure to make work output of turbines = work input to compressors and fans in gas
- ➤ Be sure to close the cycle by having (for reciprocating piston cycles) the final volume = initial volume or (for propulsion cycles) (usually) the final pressure = ambient pressure

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